



# ELEC 341: Systems and Control

## Lecture 10

### Time response: Examples

# Course roadmap

## Modeling

- ✓ Laplace transform
- ✓ Transfer function
- Models for systems
  - ✓ • Electrical
  - ✓ • Electromechanical
  - ✓ • Mechanical
- ✓ Linearization, delay

## Analysis

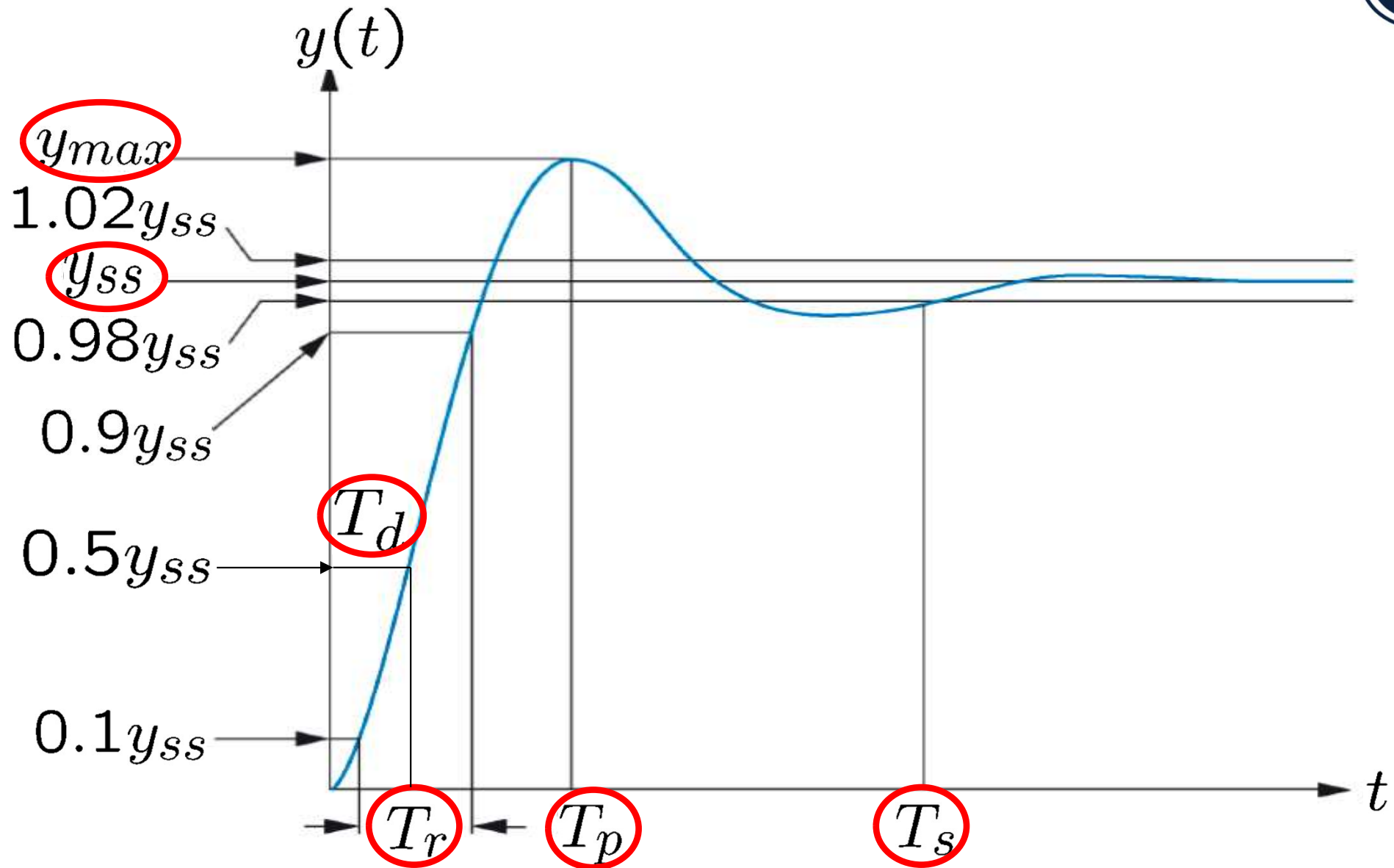
- ✓ Stability
  - ✓ • Routh-Hurwitz
  - Nyquist
- Time response
  - ✓ • Transient
  - ✓ • Steady state
- Frequency response
  - Bode plot

## Design

- Design specs
- Root locus
- Frequency domain
- PID & Lead-lag
- Design examples

*Matlab simulations*

# Typical step response (review)



# Performance measures

- Transient response

- Peak value
- Peak time
- Percent overshoot
- Delay time
- Rise time
- Settling time

(Done for 1<sup>st</sup> & 2<sup>nd</sup> order systems) ✓

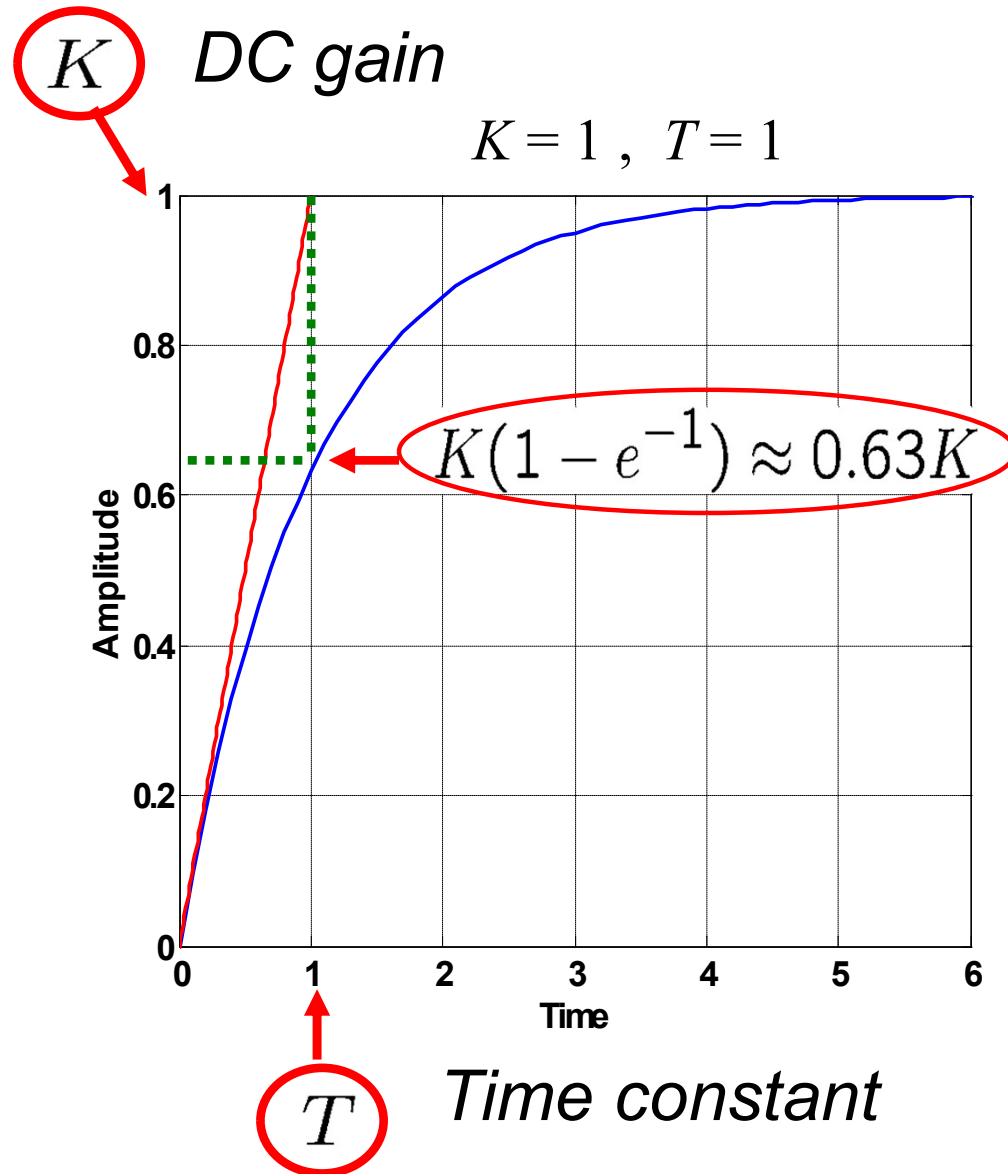
- Steady state response

- Steady state error

*We connected  
these performance  
measures  
with s-domain.* ✓

(Done) ✓

# Step response for 1<sup>st</sup>-order system (review)



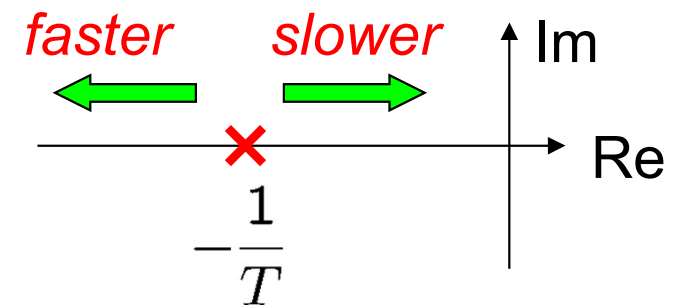
$$G(s) = \frac{K}{Ts + 1}$$

Settling time

$$5\%T_s : 3T = \frac{3}{|\text{Re}|}$$

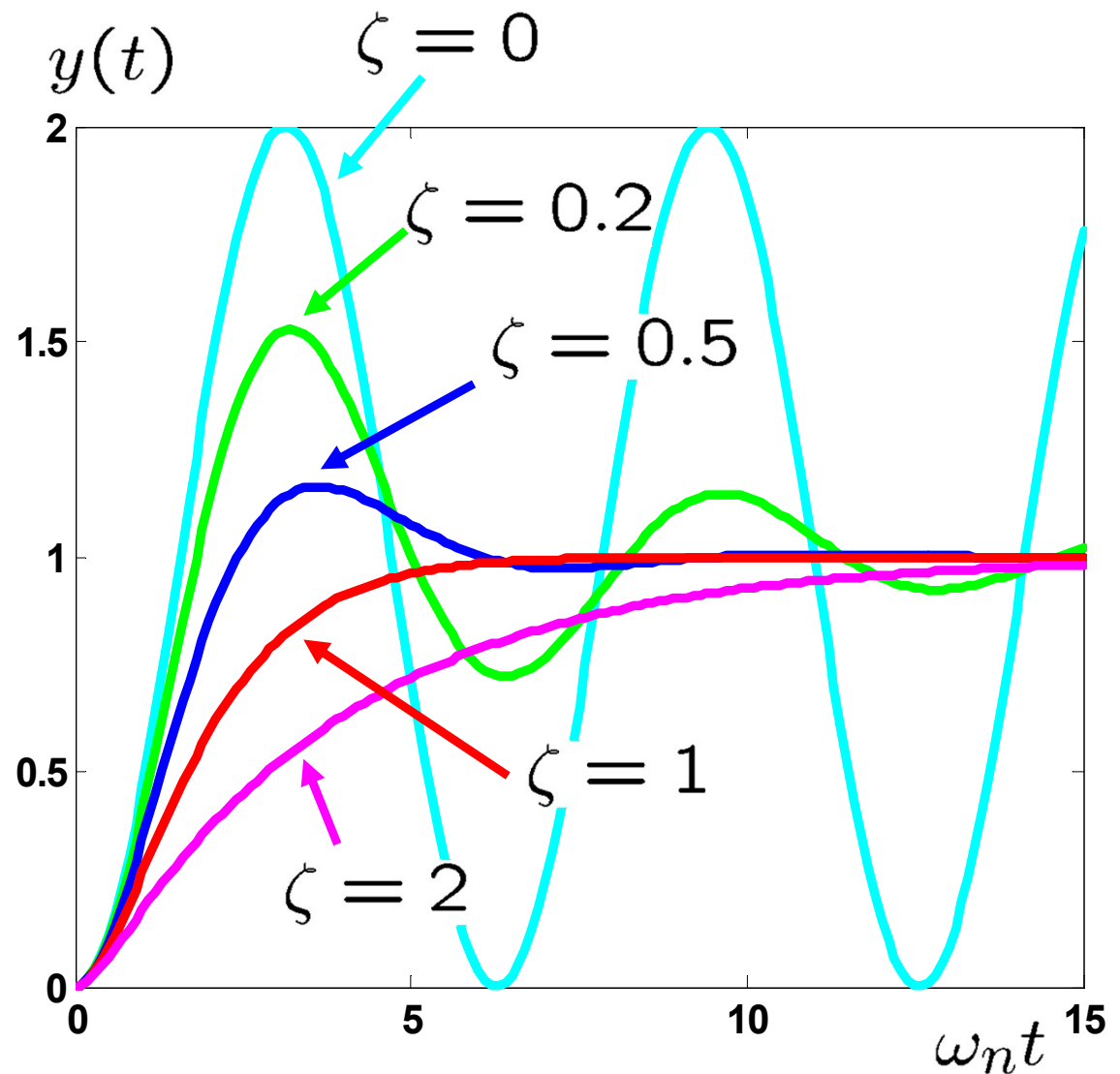
$$2\%T_s : 4T = \frac{4}{|\text{Re}|}$$

✗ is used to show poles



## Step response of 2<sup>nd</sup>-order system for various damping ratios (review)

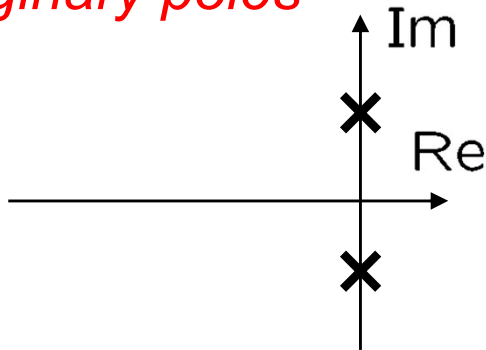
- Undamped  
 $\zeta = 0$
- Underdamped  
 $0 < \zeta < 1$
- Critically damped  
 $\zeta = 1$
- Overdamped  
 $\zeta > 1$



# Pole locations & damping (2<sup>nd</sup> order Systems)

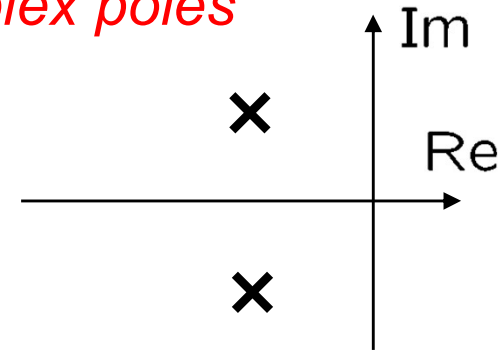
- Undamped

*Imaginary poles*



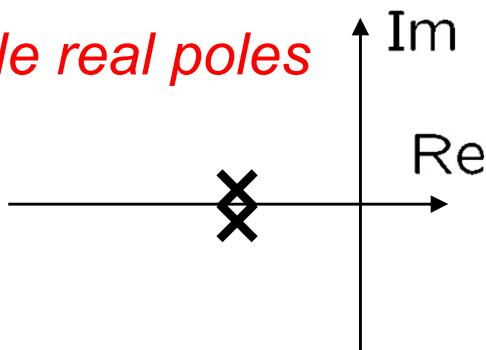
- Underdamped

*Complex poles*



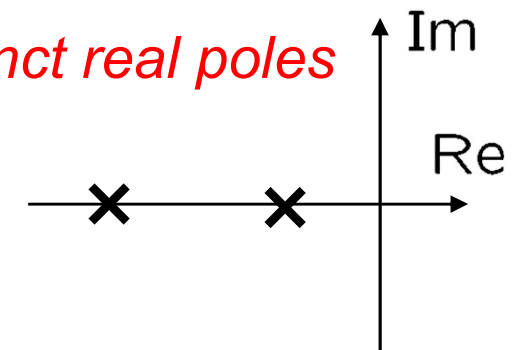
- Critically damped

*Double real poles*



- Overdamped

*Distinct real poles*



# Step response for 2<sup>nd</sup> order system: Underdamped case (review)

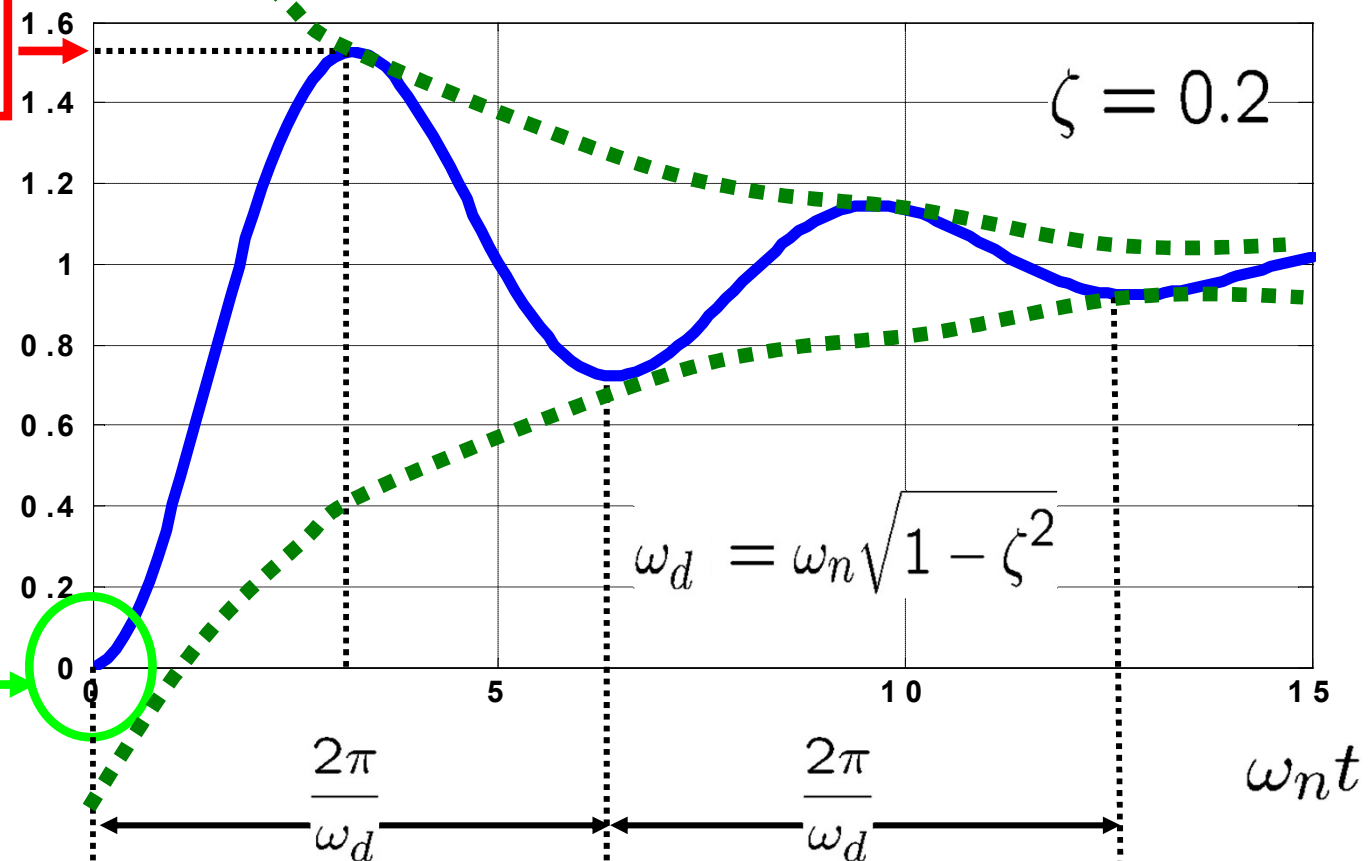
$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2} \quad 0 < \zeta < 1$$

$$y_{max} = 1 + e^{-\frac{\zeta\pi}{\sqrt{1-\zeta^2}}}$$

$$T_p = \frac{\pi}{\omega_d}$$

$$y(0) = 0$$

$$y'(0) = 0$$





# Properties of underdamped 2<sup>nd</sup> order system in terms of $\zeta$ and $\omega_n$ (review)

$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2} \quad 0 < \zeta < 1$$

	(5%)	(2%)
Settling time, $T_s$	$\approx \frac{3}{\zeta\omega_n}$ or $\frac{4}{\zeta\omega_n}$	
Peak time, $T_p$	$\frac{\pi}{\omega_d} = \frac{\pi}{\omega_n \sqrt{1-\zeta^2}}$	
Peak value, $y_{max}$	$1 + e^{-\zeta\pi/\sqrt{1-\zeta^2}}$	
Percent overshoot, $PO$ or $\%OS$	$100e^{-\zeta\pi/\sqrt{1-\zeta^2}}$	

Time constant =  $T = \frac{1}{\zeta\omega_n}$

Delay time =  $T_d = \frac{1 + 0.7\zeta}{\omega_n}$

$PO = 100e^{-\pi/\tan\theta}$

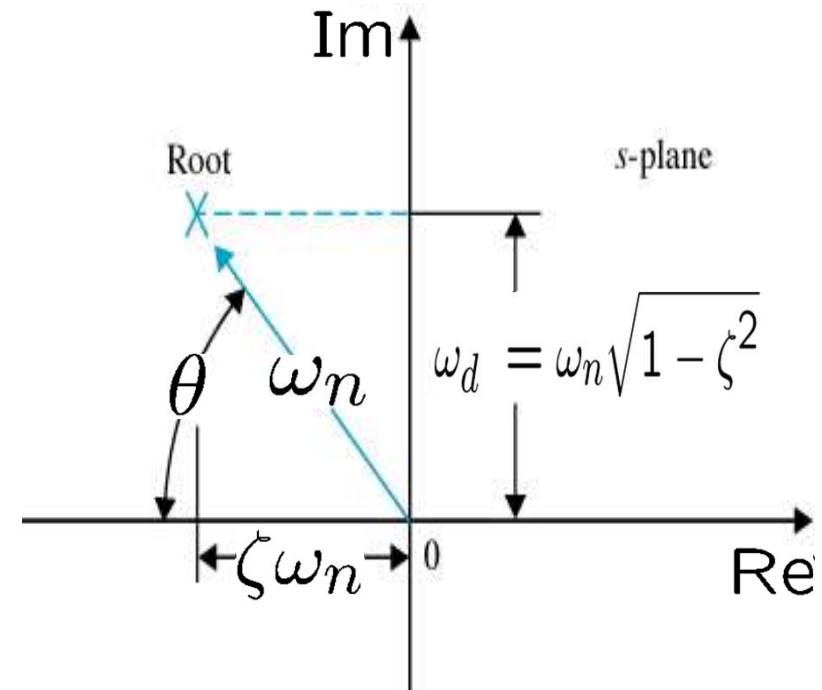
# Step response properties of underdamped 2<sup>nd</sup> order system in terms of pole locations (review)

- Poles ( $0 < \zeta < 1$ )

$$s = -\zeta\omega_n \pm j\omega_n \underbrace{\sqrt{1 - \zeta^2}}_{\omega_d}$$

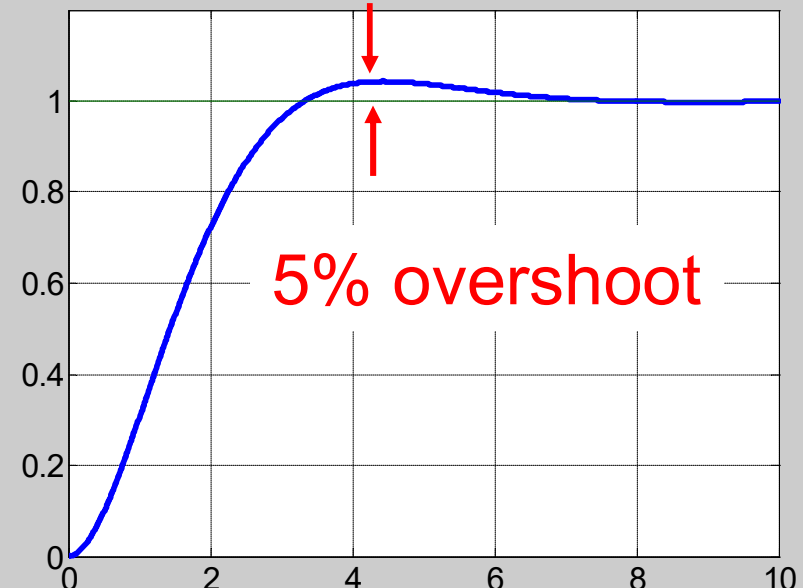
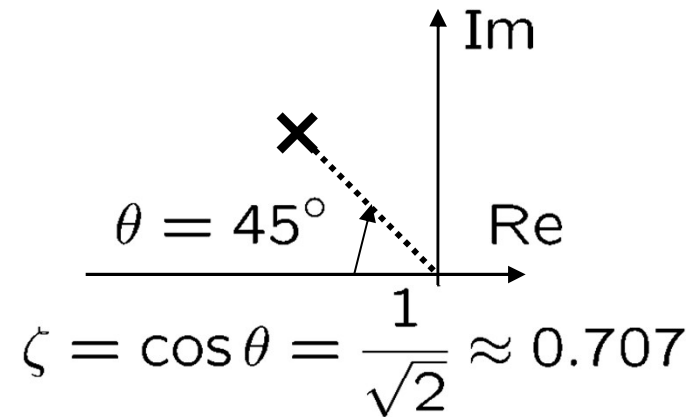
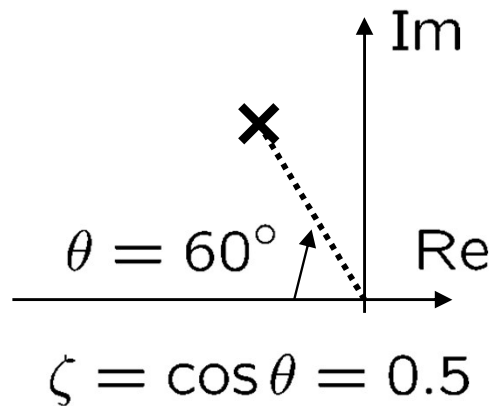
$$\zeta = \cos \theta$$

$$T = \text{Time Constant} = 1/\zeta\omega_n$$

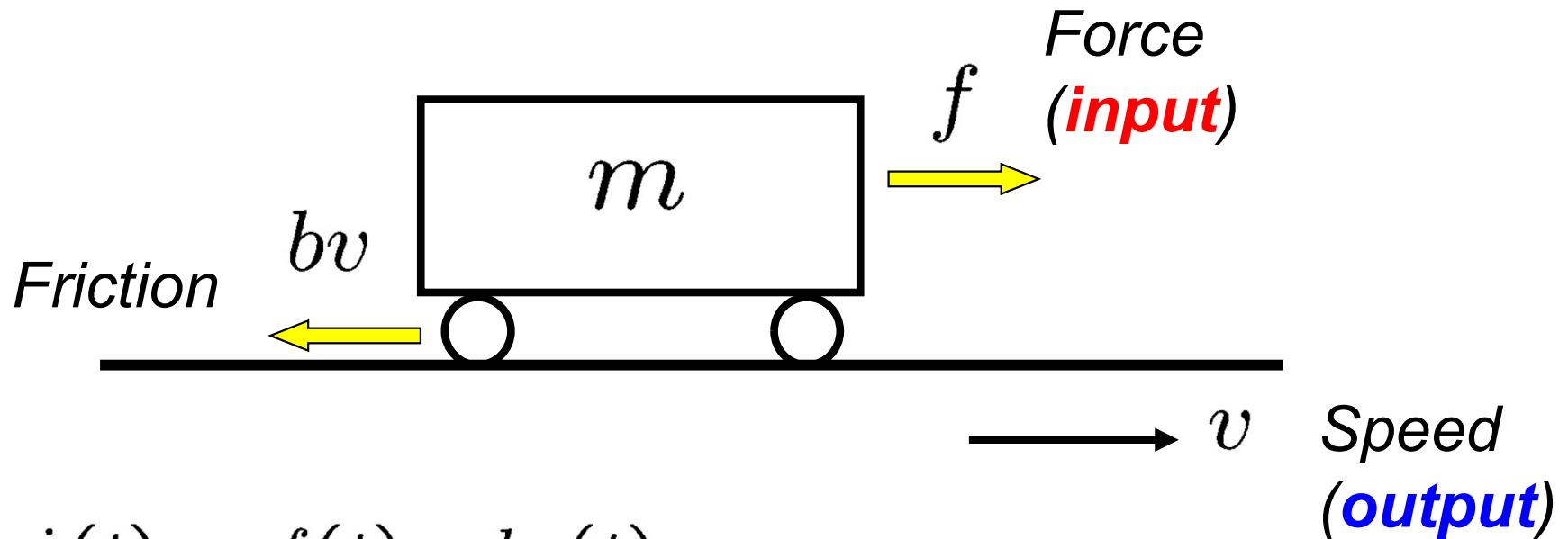


Pole			Performance	
Real part	$\zeta\omega_n$	determines	$T_s = \frac{3}{\zeta\omega_n}, \frac{4}{\zeta\omega_n}$	<div style="border: 2px solid red; padding: 10px; display: inline-block;"> <math>= \frac{3}{ \text{Re} }, \frac{4}{ \text{Re} }</math>  <math>= \frac{\pi}{ \text{Im} }</math> </div>
Imag. part	$\omega_d$	determines	$T_p = \frac{\pi}{\omega_d}$	
Angle	$\theta$	determines	overshoot	

# Angle $\theta$ and the overshoot (review)



# Example 1: Cruise control



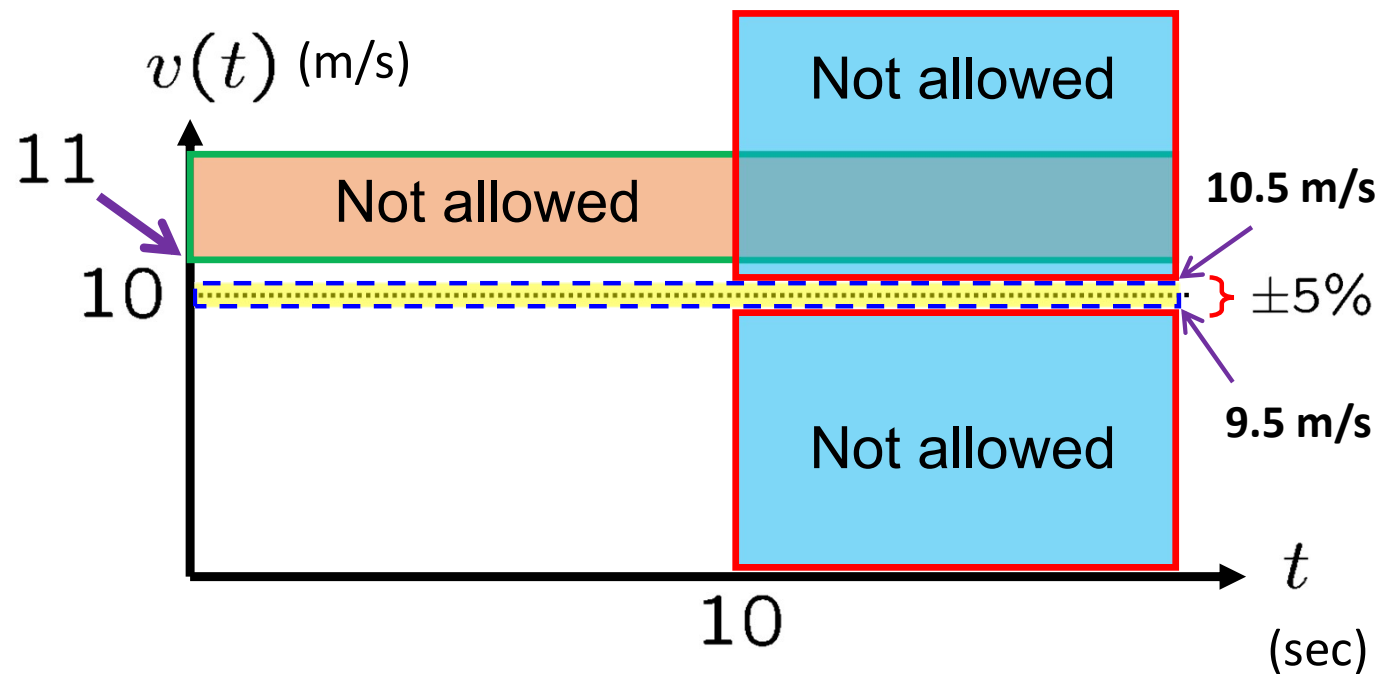
$$m\dot{v}(t) = f(t) - bv(t)$$

➔  $\frac{V(s)}{F(s)} = \frac{1}{ms + b}$

$m$	$=$	1000 (kg)
$b$	$=$	50 (N·sec/m)

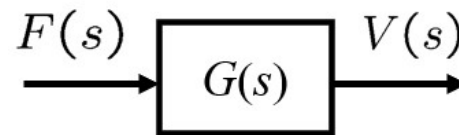
# Example 1 (cont'd)

- **Control goal:** SS (steady-state) speed 10 m/s (= 36 km/h).
  - Stable
  - Zero steady-state error
  - Percent overshoot of 10 % or less
  - $5\%T_s < 10$  sec

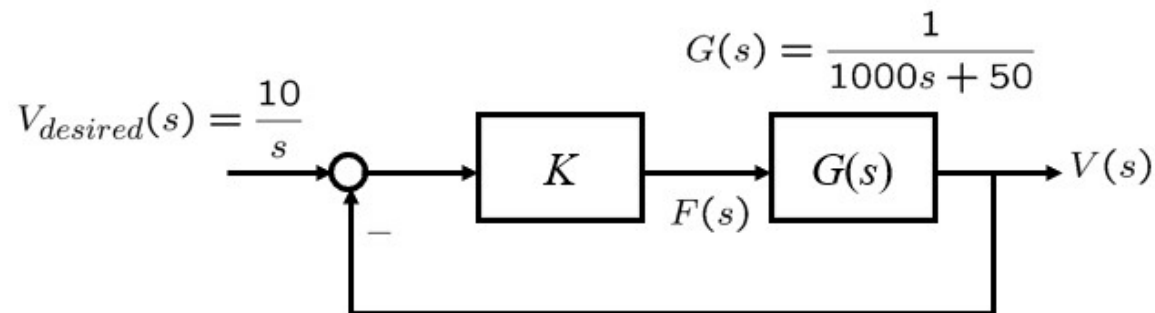


# Example 1 (cont'd)

**(a)** The following open-loop control system is given. **(a1)** Obtain the amplitude of the step force ( $R$ ) necessary to get the steady-state speed of 10 m/s. **(a2)** What is  $5\%T_s$  for this system?



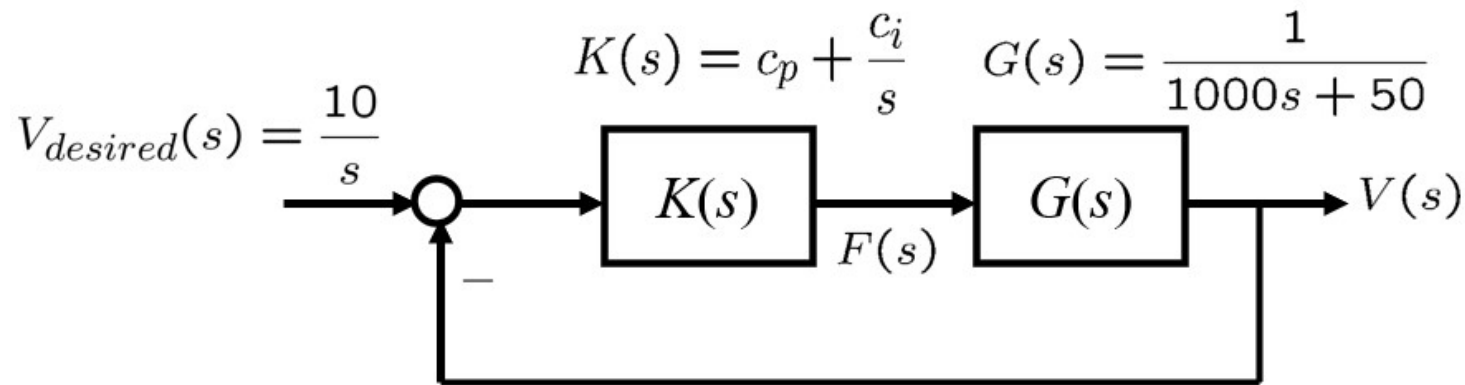
**(b)** If we decide to use a proportional controller and a negative feedback system, we will have the following block diagram:



Now apply the design requirement of  $5\%T_s < 10$  sec to the above system. **(b1)** What will be the range of  $K$ ? **(b2)** For the same block diagram, what will be the value of  $K$  if we want to satisfy the design requirement of zero steady-state error?

# Example 1 (cont'd)

(c) If we decide to use a proportional-integral controller and a negative feedback system, we will have the following block diagram:



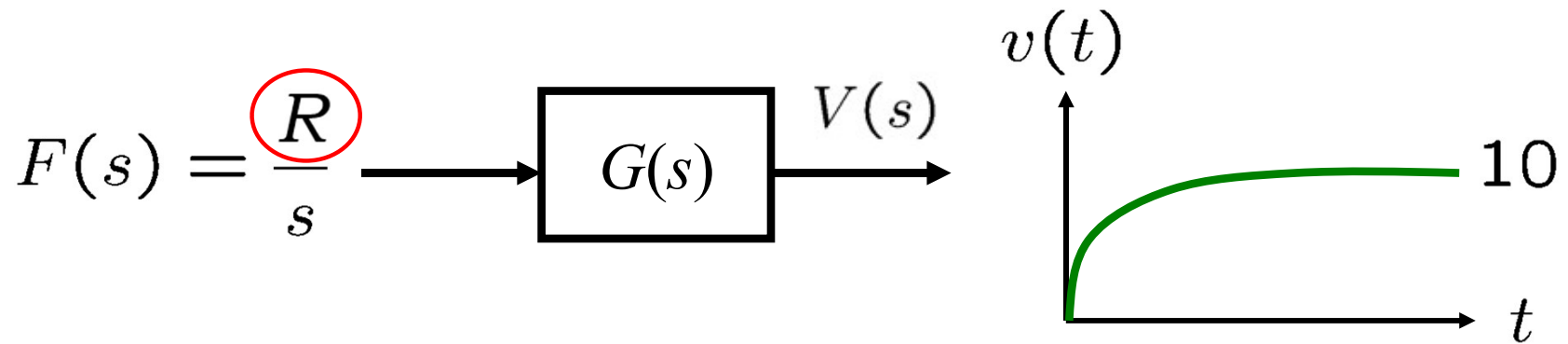
(c1) Is this system stable? (c2) What is the value of  $K_p$  (i.e., position error constant)?  
(c3) What is its steady-state error? (c4) Without finding  $c_p$  and  $c_i$ , discuss how one might compute the numerical values of these parameters?

## Example 1: Open-loop control (cont'd)

**Solution:**

(a1) 
$$G(s) = \frac{V(s)}{F(s)} = \frac{1}{1000s + 50} = \frac{1/50}{20s + 1} \rightarrow T = 20$$

- Obtain the **amplitude** of the step force necessary to get the steady-state speed of 10 m/s. This is shown by  $R$ .



$$v_{ss} = R G(0) = 10 \Rightarrow R = \frac{10}{G(0)} = 500(N) \Rightarrow \boxed{R = 500}$$

(a2)

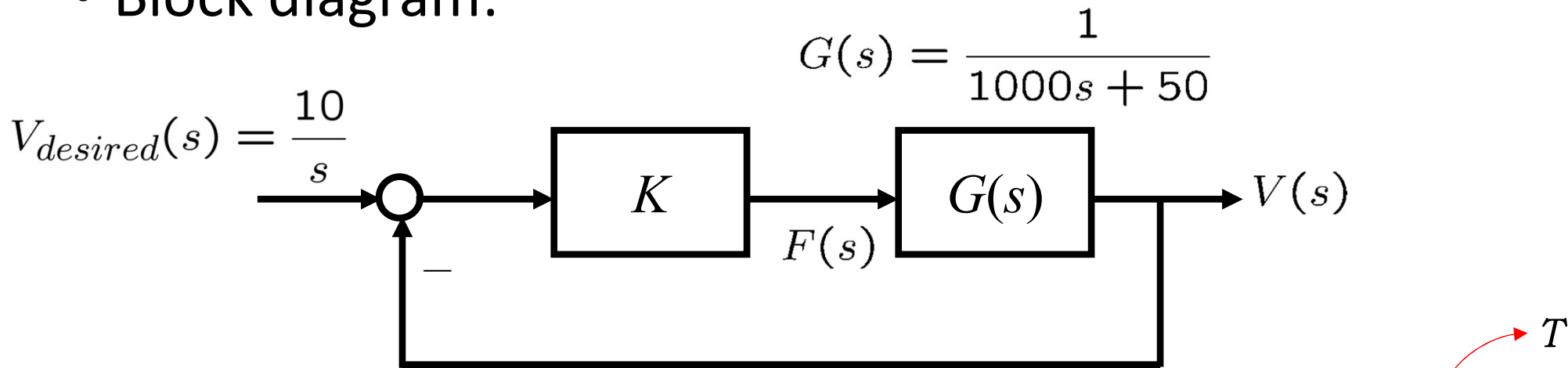
- Note that  $5\%T_s = 3T$  or  $3 \times 20 = 60$  sec. However, we need it to be less than 10 sec.
- So, by open-loop control,  $5\%T_s = 60$  sec! and thus **Too Slow!**



# Example 1: P control (cont'd)

(b1)

- Block diagram:



$$\frac{V(s)}{V_{desired}(s)} = \frac{KG(s)}{1 + KG(s)} = \frac{K}{1000s + 50 + K} = \frac{K/(50 + K)}{[1000/(50 + K)]s + 1}$$

*No overshoot*

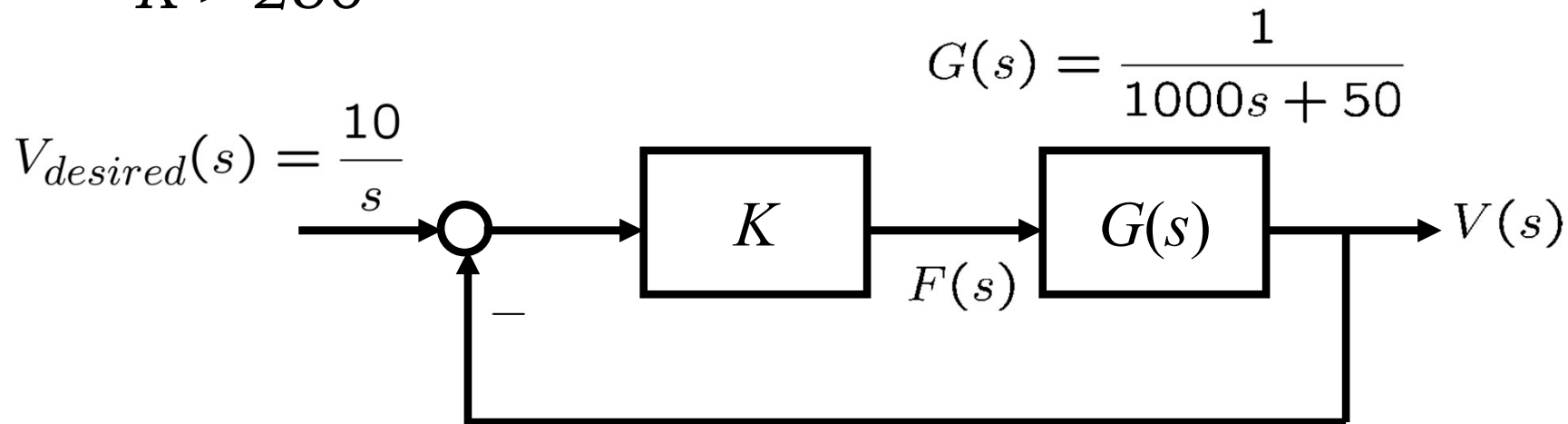
- $5\%T_s < 10 \text{ sec} \Rightarrow 3T < 10 \text{ sec}$

$$\Rightarrow 3 \times \frac{1000}{50 + K} < 10 \Rightarrow 300 < 50 + K \Rightarrow K > 250$$

# Example 1: P control (cont'd)

(b2)

- $K > 250$



- To have zero SS error:

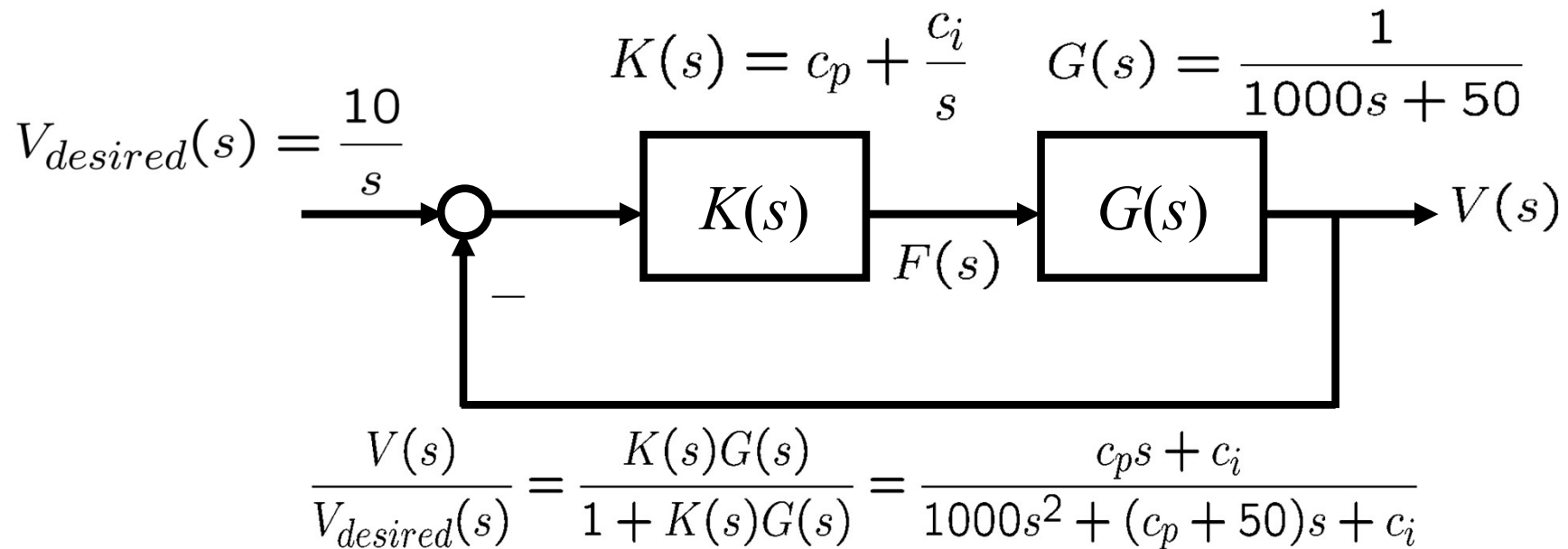
$$e_{ss} = \frac{10}{1 + K_p} = \frac{10}{1 + L(0)} = \frac{10}{1 + KG(0)} = \frac{10}{1 + K/50}$$

$$e_{ss} = 0 \Rightarrow K = \infty$$

# Example 1: PI control (cont'd)

(c1)

- Block diagram:



- Stable: **OK**

(c2)

$$K_p = L(0) = K(0)G(0) = \infty \quad \Rightarrow \quad K_p = \infty$$

(c3)

- Zero SS error: **OK**

**Note:**  $e_{ss} = R/(1+K_p) = 10/(1+\infty) = 0$

# Example 1: PI control (cont'd)

(c4)

- How to satisfy transient requirements?
  - **Tune controller parameters** to place poles of the closed-loop transfer function at the “right places”.

$$\frac{V(s)}{V_{desired}(s)} = \frac{c_p s + c_i}{1000s^2 + (c_p + 50)s + c_i}$$

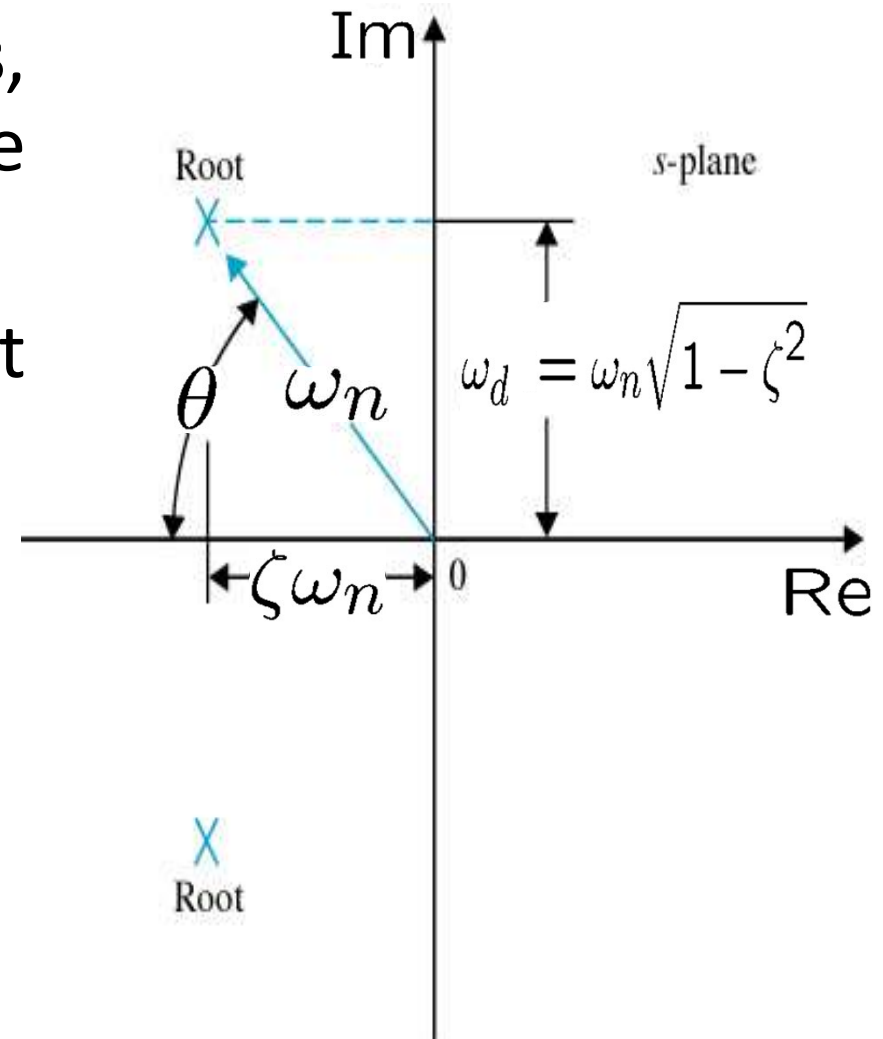
- We will learn
  - where the “right places” are (today’s lecture: Example 2)
  - how to tune controller parameters (from next lecture)
- We can show (by trial and error and computer simulation) that the following controller is one of the controllers that can satisfy all the required conditions (not the topic of this lecture):

$$K(s) = 1050 + \frac{274}{s}$$

## Example 2

- For the following **design specs**, find the **allowable region** of the second-order poles:

$PO$  at most 5% and  $T_s$  (5%) at most 7 sec.



**Note:**

**Design Specs** (i.e., Design Specifications) is just another term for **Performance Measures**.

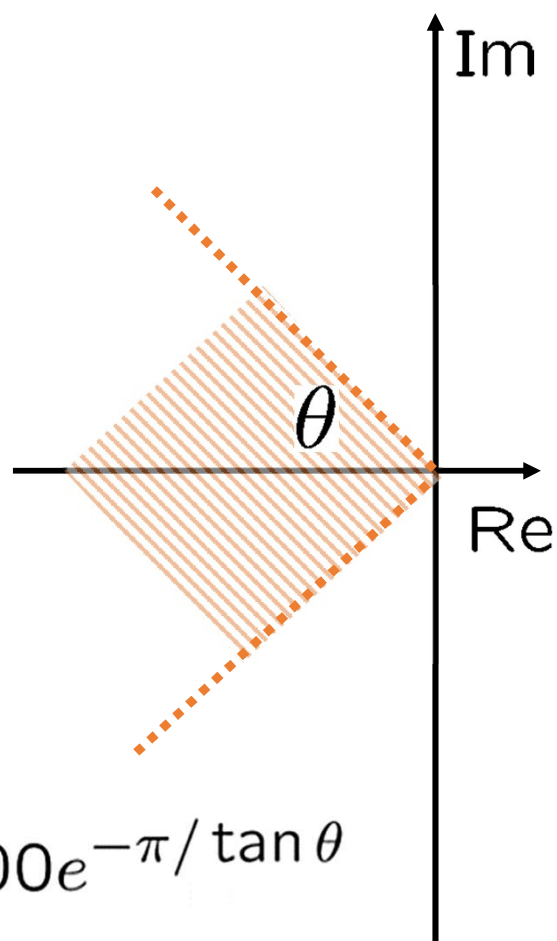
## Example 2 (cont'd)

### Solution:

- From the given percent overshoot constraint, **PO constraint**, we can obtain **allowable angles  $\theta$** .
- From the given settling time constraint,  **$T_s$  constraint**, we can obtain **allowable real part of poles**.

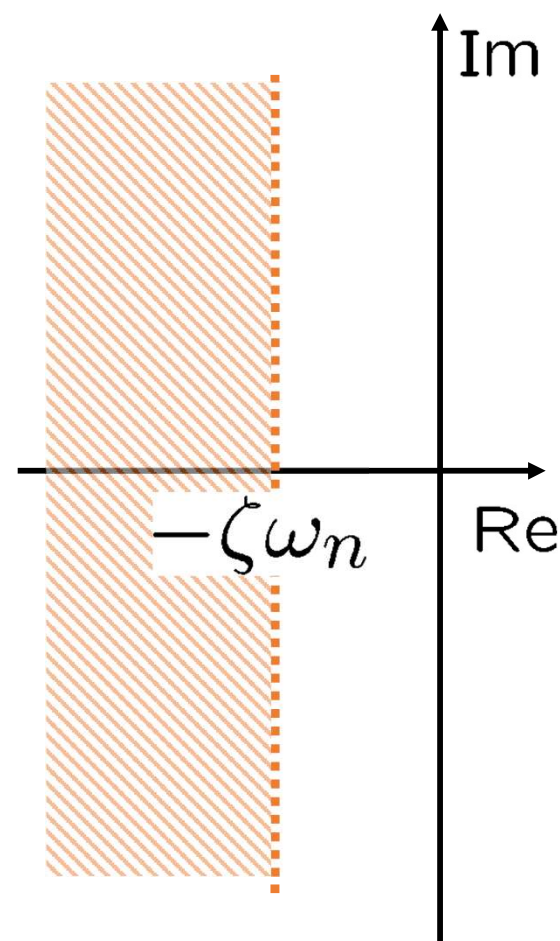
# Example 2 (cont'd)

**$PO$  constraint**



$$PO = 100e^{-\pi/\tan \theta}$$

**$T_s$  constraint**

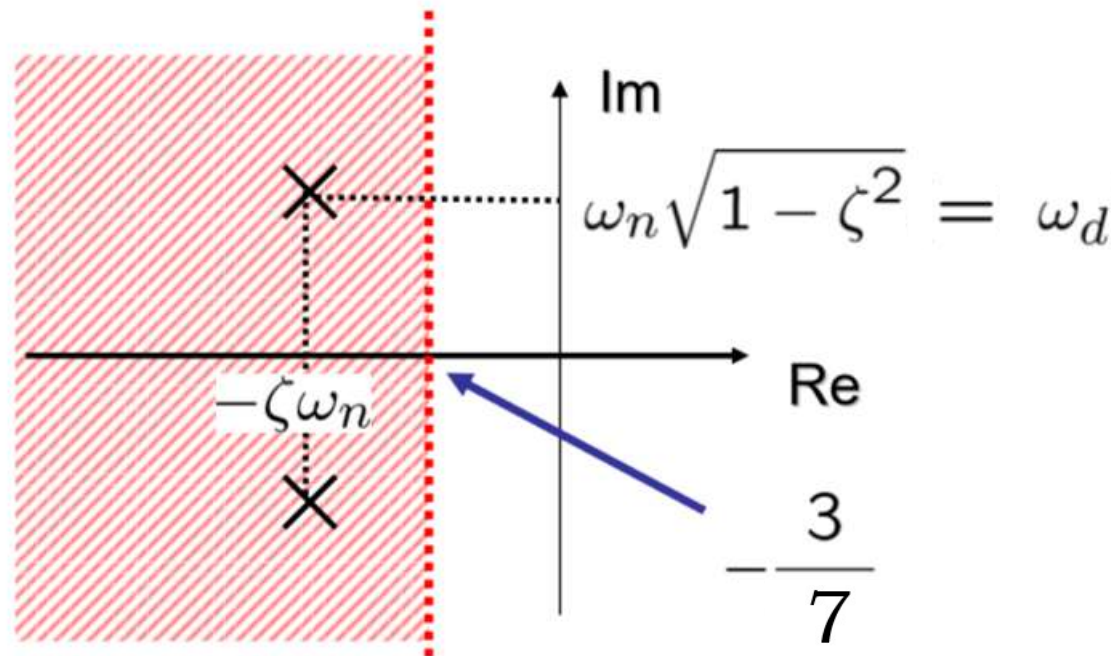


$$T_s = \frac{3}{\zeta\omega_n}$$

## Example 2 (cont'd)

- Require 5% settling time  $T_s < 7$

$$T_s = \frac{3}{\zeta\omega_n} \rightarrow \frac{3}{\zeta\omega_n} < 7 \rightarrow \boxed{\zeta\omega_n > \frac{3}{7}}$$





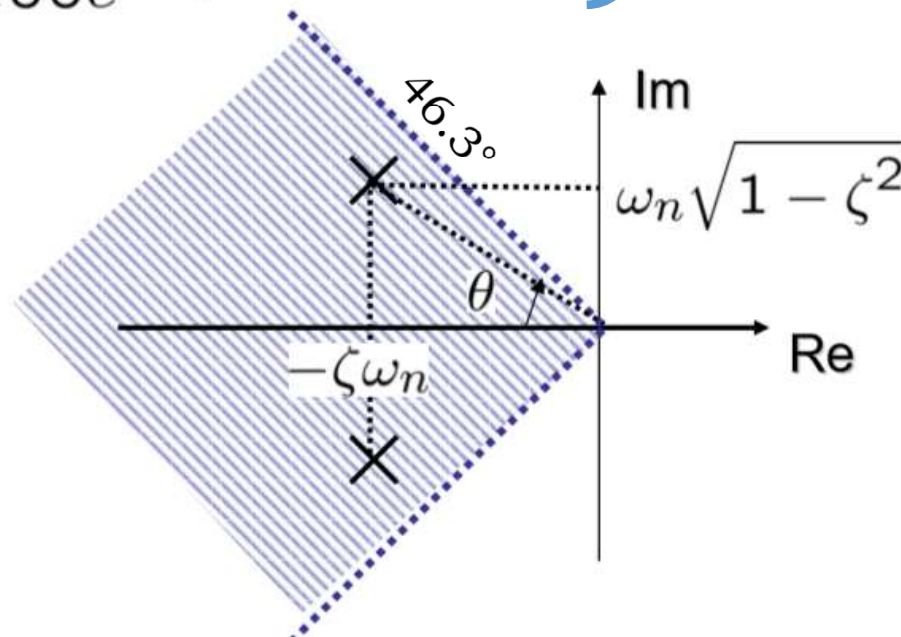
# Example 2 (cont'd)

- Require  $PO < 5\%$

$$PO = 100e^{-\zeta\pi/\sqrt{1-\zeta^2}} < 5\%$$

$$= 100e^{-\pi/\tan\theta}$$

$$\theta < 46.3^\circ$$

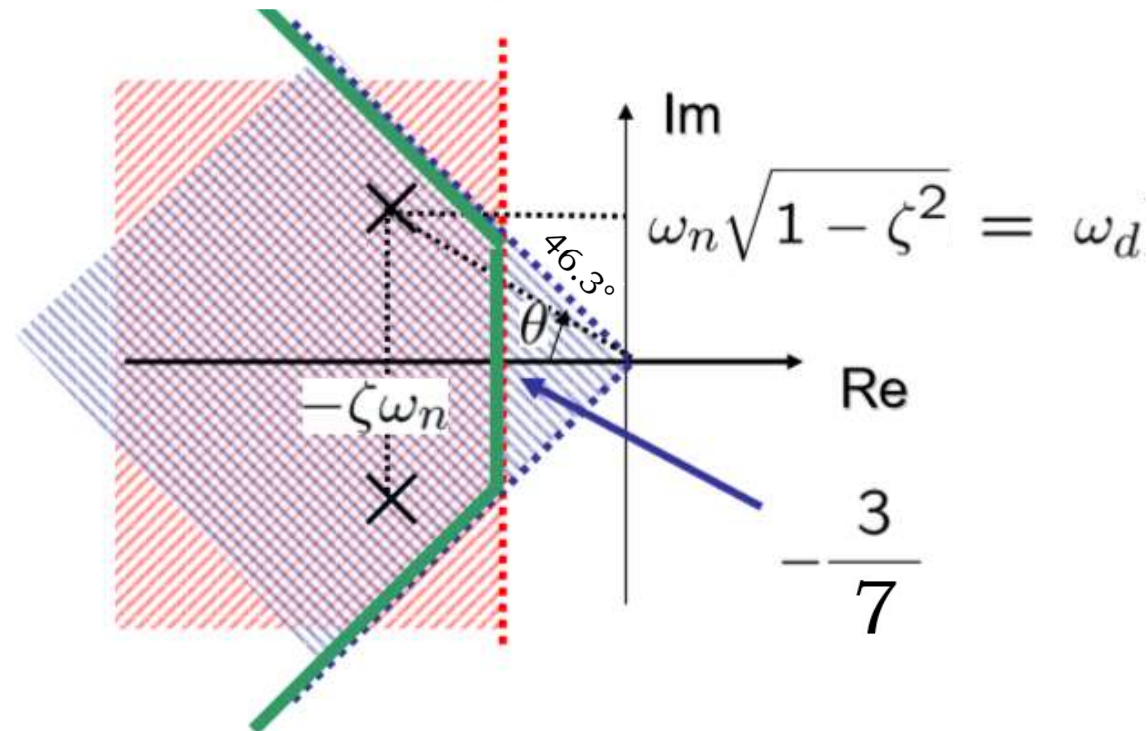


$PO_{max}$	$\theta_{max}$
5%	46.3°

## Example 2 (cont'd)

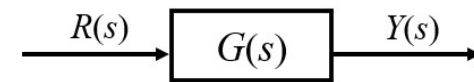
- Combination of two requirements

$$\zeta\omega_n > \frac{3}{7} \quad \& \quad \theta < 46.3^\circ$$



# Example 3

- Consider a 2<sup>nd</sup> order **overdamped** system



$$G(s) = \frac{10}{s^2 + 11s + 10} = \frac{10}{(s + 1)(s + 10)}$$

For unit step input, obtain (or estimate)

- steady state value,
- percent overshoot, and
- 2% settling time.

**Important Note:** The formulas for 2<sup>nd</sup> order system that were summarized in a table can only be used when  $0 < \zeta < 1$ . That is, if  $\zeta$  is greater than 1, we cannot use the formulas in the table.

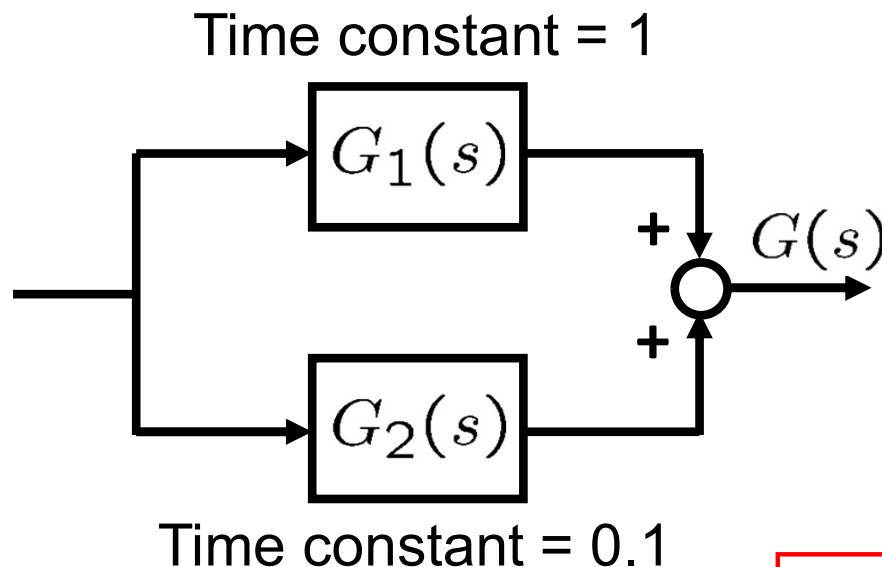
**Solution:**

- Steady state value is the DC gain  $G(0) = 1$ .
- No overshoot (overdamped!)

## Example 3 (cont'd)

- Poles are -1 and -10.

$$G(s) = \frac{10}{s^2 + 11s + 10} = \underbrace{\frac{A}{s+1}}_{G_1(s)} + \underbrace{\frac{B}{s+10}}_{G_2(s)}$$



*Time constant of  $G(s)$  is estimated by the slower subsystem  $G_1(s)$ , i.e., the **larger  $T$** .*

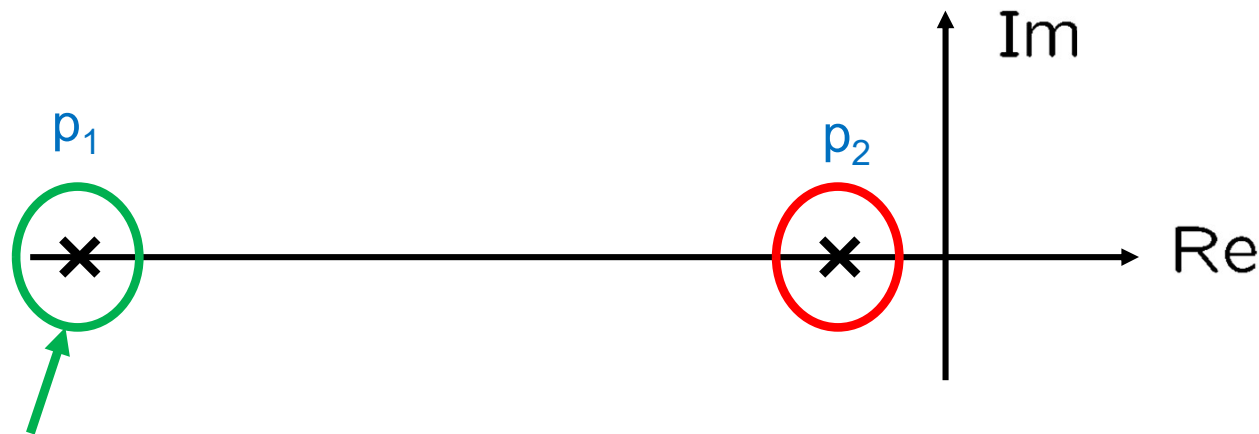
$$2\%T_s = 4T. \text{ In this case } (T=1). \text{ So, } 2\%T_s \approx 4 \times 1 \approx 4 \text{ sec}$$



2% settling time: ***about* 4 seconds**


## Example 3 (cont'd)

- Dominant poles:** Poles closest to the imaginary axis and far away from remaining poles dominate the behavior of responses. We only compare the real parts of the roots when determining dominance because it is the real part that determines how fast the response decreases.



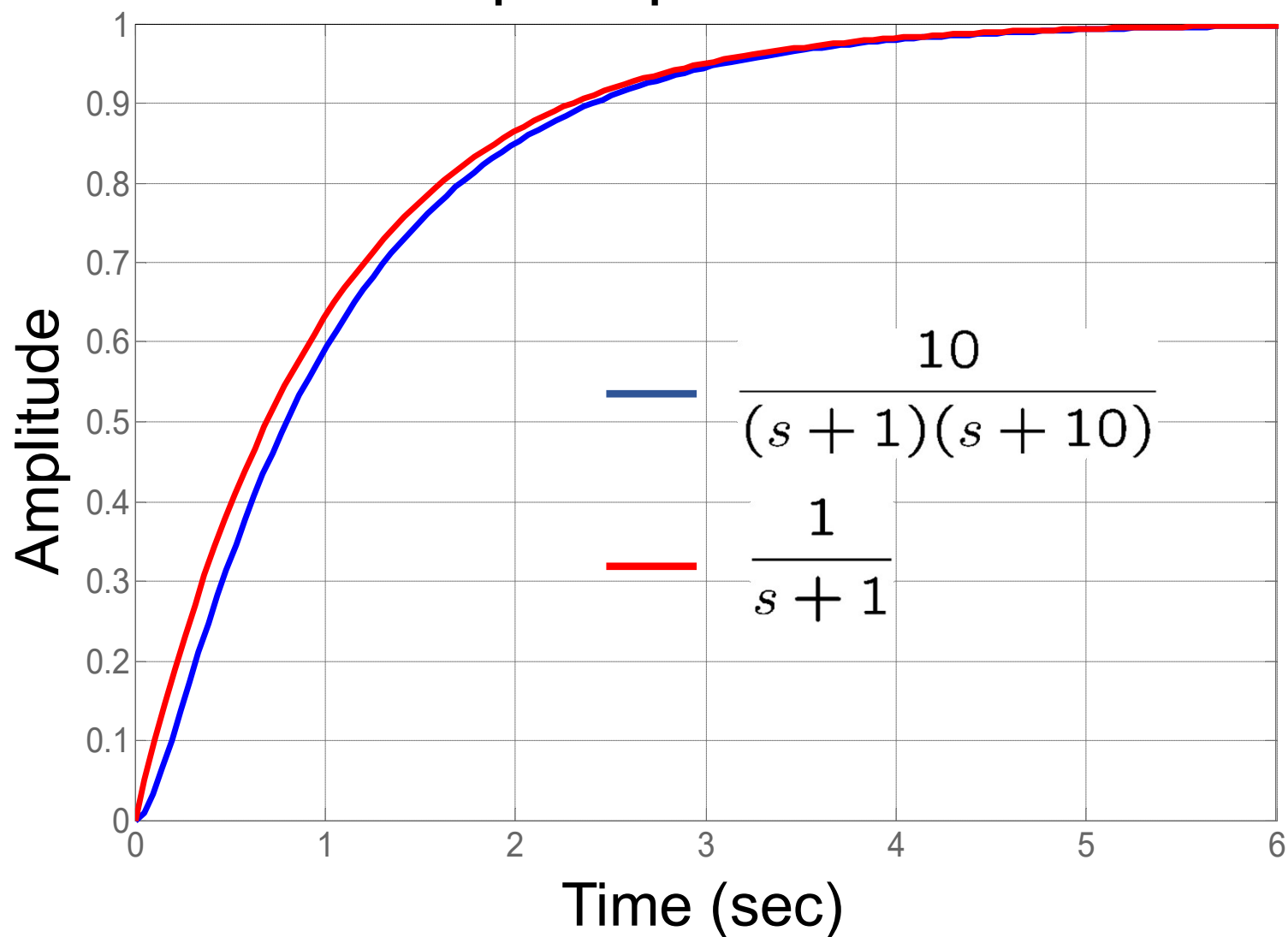
Poles far left (5-10 times) from dominant poles may be ignored. That is, if  $p_1 > 5p_2$  or  $10p_2$ , we can ignore the effect of  $p_1$ .

$$G(s) = \frac{10}{(s+1)(s+10)} \approx \frac{1}{s+1}$$


 Same DC gain

## Example 3 (cont'd)

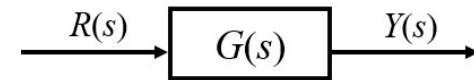
### Step responses



# Example 4

- For the unit step response of a stable 3<sup>rd</sup>-order system,

$$G(s) = \frac{8}{(s + 2.5)(s^2 + 2s + 4)}$$



obtain (or estimate)

- steady state value,
- 2% settling time, and
- percent overshoot.

**Solution:**

- Steady state value is the DC gain  $G(0) = 0.8$ .

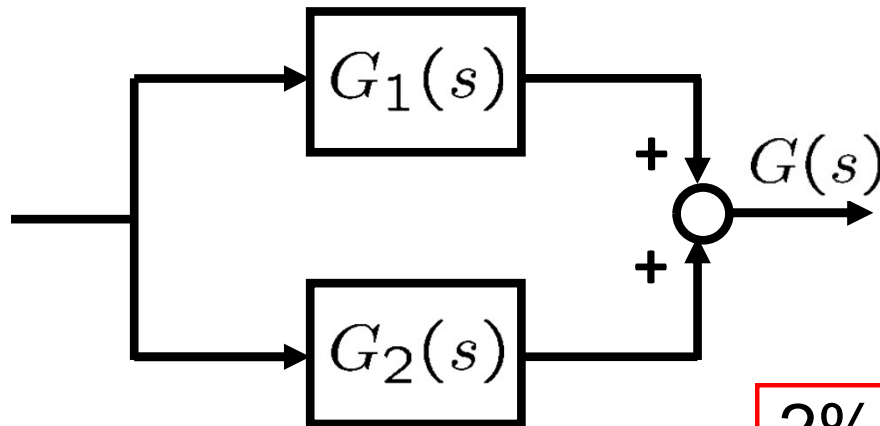
# Example 4 (cont'd)

- Partial fraction expansion

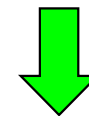
$$G(s) = \frac{8}{(s + 2.5)(s^2 + 2s + 4)} = \underbrace{\frac{A}{s + 2.5}}_{G_1(s)} + \underbrace{\frac{Bs + C}{s^2 + 2s + 4}}_{G_2(s)}$$

$$\omega_n = 2, \zeta = 0.5$$

Time constant =  $1/2.5 = 0.4$  s



*Time constant of  $G(s)$  is estimated by the slower subsystem  $G_2(s)$ , i.e., the **larger  $T$** .*



2% settling time: **about 4 seconds**

Time constant =  $1/(\zeta\omega_n) = 1$  s

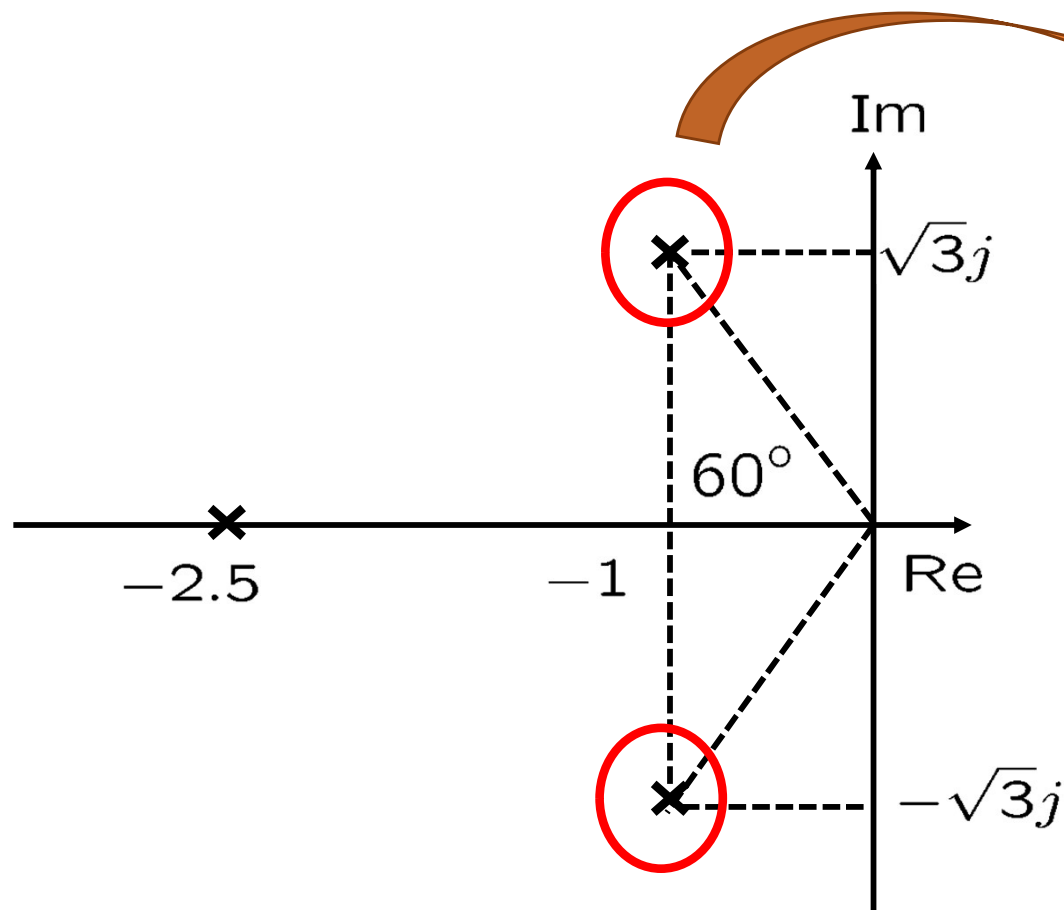
$$PO = 100e^{-\zeta\pi/\sqrt{1-\zeta^2}} \quad \zeta = 0.5$$

$PO = 16.3\%$



## Example 4 (cont'd)

- Poles  $s = -2.5, -1 \pm \sqrt{3}j$

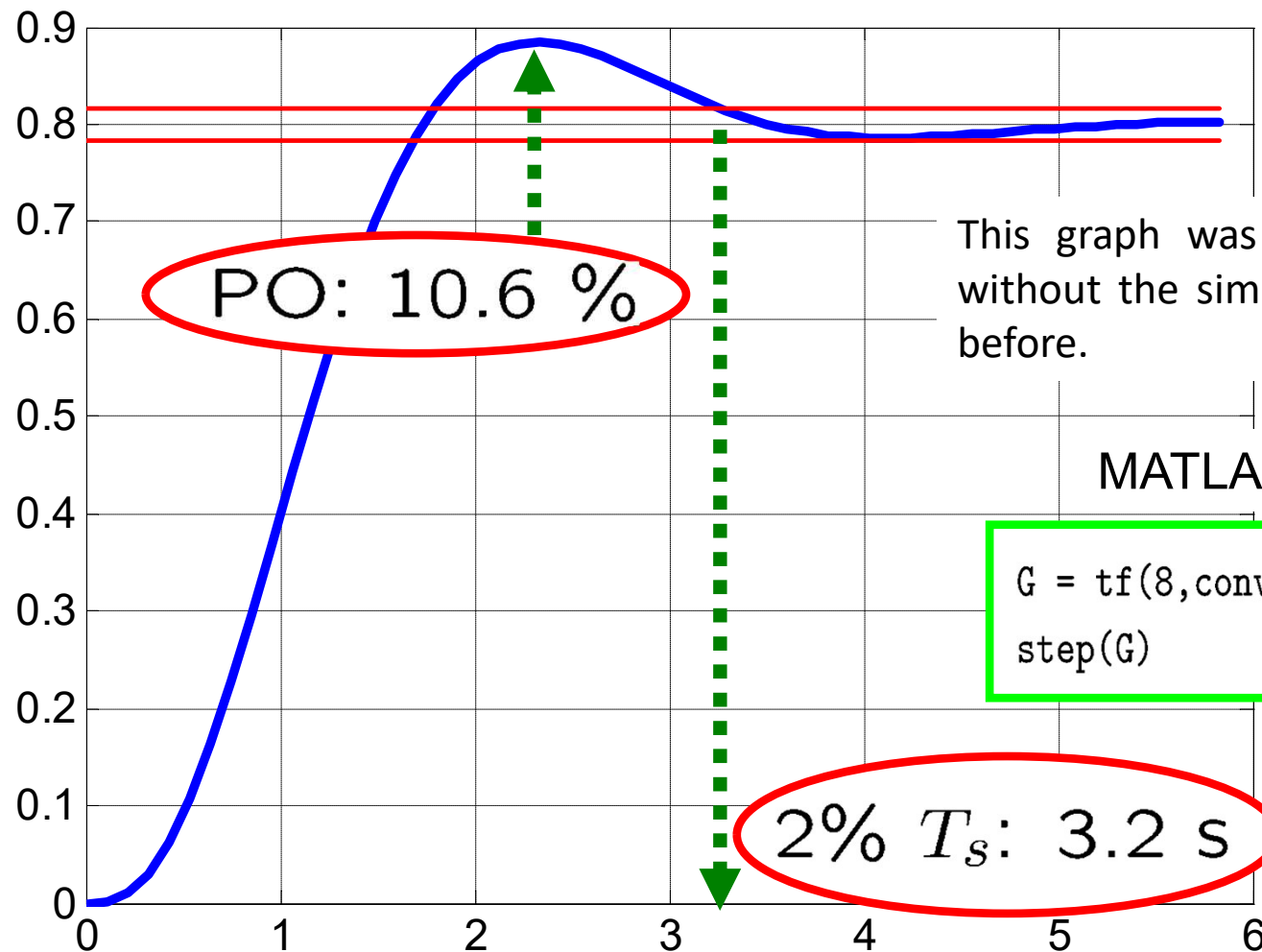


*These poles are not  
“sufficiently dominant”!*

*Estimation based on these  
poles may be inaccurate.*

# Example 4 (cont'd)

## Step response of $G(s)$



This graph was obtained accurately without the simplification mentioned before.

MATLAB command

```
G = tf(8,conv([1 2.5],[1 2 4]));  
step(G)
```

# Example 5

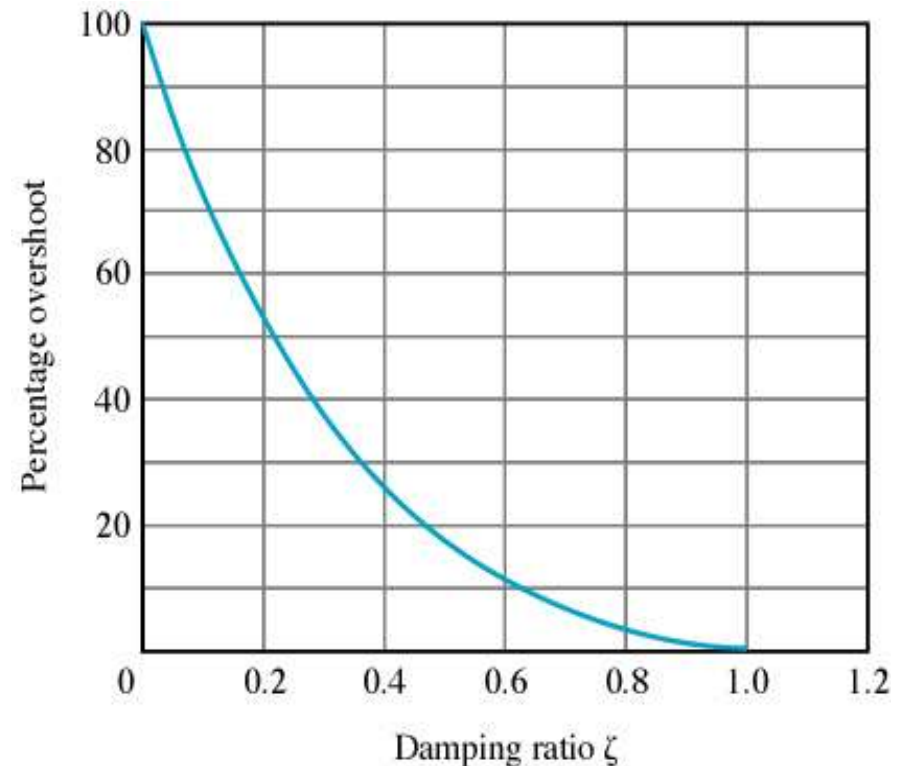
- For the standard 2<sup>nd</sup> order systems, derive the relationship for damping ratio as a function of percent overshoot, from the following formula:

$$PO = 100e^{-\zeta\pi/\sqrt{1-\zeta^2}}$$

**Solution:**

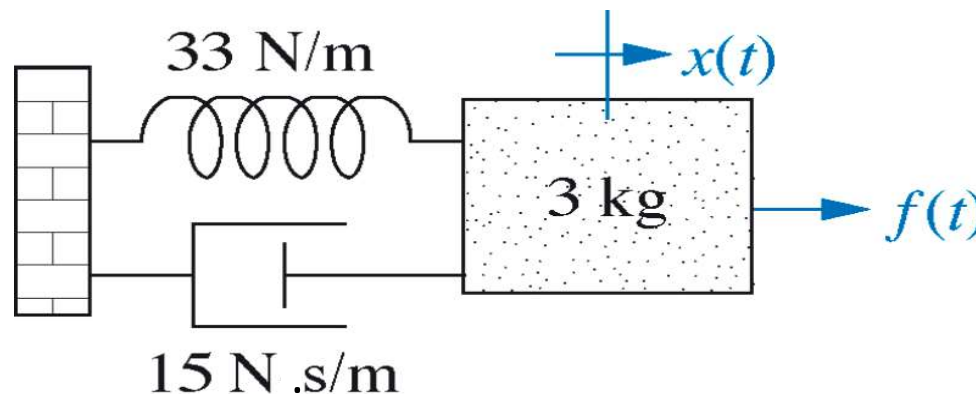
$$\left(\ln \frac{PO}{100}\right)^2 = \frac{\zeta^2\pi^2}{1-\zeta^2}$$
$$\Leftrightarrow \zeta^2 = \frac{\left(\ln \frac{PO}{100}\right)^2}{\pi^2 + \left(\ln \frac{PO}{100}\right)^2}$$

$$\Leftrightarrow \zeta = \frac{\left|\ln \frac{PO}{100}\right|}{\sqrt{\pi^2 + \left(\ln \frac{PO}{100}\right)^2}}$$



# Example 6

- For the system below,
  - Find the transfer function from  $F(s)$  to  $X(s)$ , i.e.,  $F(s)$  is the input and  $X(s)$  is the output.
  - Find DC gain,  $\zeta$ ,  $\omega_n$ ,  $\omega_d$ ,  $T_s$  (2%),  $T_p$ , and PO.



**Solution:**

$$G(s) = \frac{X(s)}{F(s)} = \frac{1}{3s^2 + 15s + 33}$$

## Example 6 (cont'd)

- Find DC gain  $G(0)$ ,  $\zeta$ ,  $\omega_n$ ,  $\omega_d$ ,  $T_s$  (2%),  $T_p$ , and PO.

$$G(s) = \frac{1}{3s^2 + 15s + 33} = \frac{1}{33} \cdot \frac{11}{s^2 + 5s + 11} \quad 2\zeta\omega_n = 5, \quad \omega_n^2 = 11$$

$$G(0) = 1/33 = 0.03$$

$$\omega_n = \sqrt{11} = 3.31 \quad ; \quad \zeta = \frac{5}{2\sqrt{11}} = 0.75 \quad ; \quad \omega_d = \omega_n \sqrt{1 - \zeta^2} = \frac{\sqrt{19}}{2} = 2.17$$

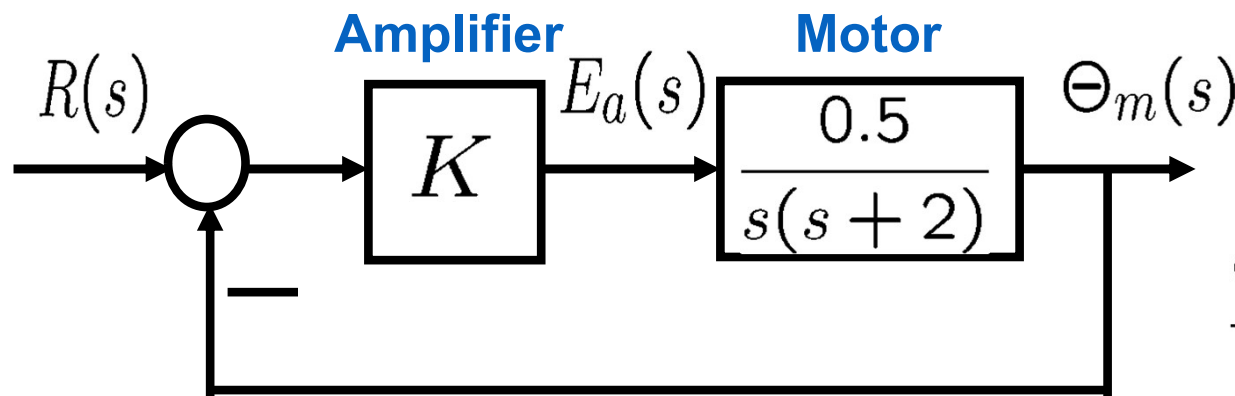
$$T_s = \frac{4}{\zeta\omega_n} = 1.6 \quad ; \quad T_p = \frac{\pi}{\omega_d} = \frac{2\pi}{\sqrt{19}} = 1.44 \quad ; \quad PO = 100e^{-\zeta\pi/\sqrt{1-\zeta^2}} = 2.72\%$$



$$\begin{aligned} G(0) &= 0.03 \\ \zeta &= 0.75 \\ \omega_n &= 3.31 \\ \omega_d &= 2.17 \\ T_s &= 1.6 \\ T_p &= 1.44 \\ PO &= 2.72\% \end{aligned}$$

# Example 7

- DC motor position control



Closed-loop TF

$$\frac{\Theta_m(s)}{R(s)} = \frac{0.5K}{s^2 + 2s + 0.5K}$$

- Design the amplifier gain  $K$  so that
  - Percent overshoot is 5%.

$$\begin{cases} 2\zeta\omega_n = 2 \\ \omega_n^2 = 0.5K \end{cases}$$

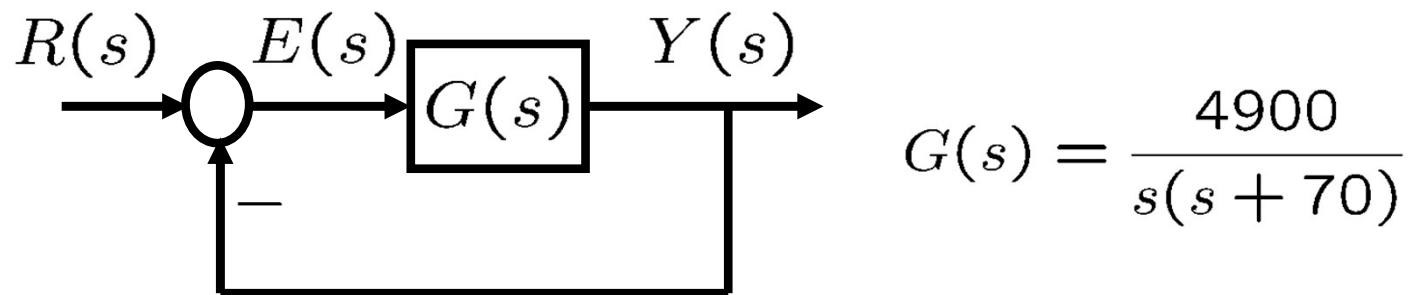
**Solution:**

$$\zeta = \frac{\left| \ln \frac{PO}{100} \right|}{\sqrt{\pi^2 + \left( \ln \frac{PO}{100} \right)^2}} = \frac{1}{\sqrt{2}} \quad \longrightarrow \quad \omega_n = \sqrt{2} \quad \longrightarrow \quad K = \frac{\omega_n^2}{0.5}$$

$$K = 4$$

# Example 8

- For the unity feedback system, check and compute the followings:



- (a) Stability
- (b) 2% settling time for  $r(t) = u(t)$  (unit step input)
- (c) Steady-state errors:
  - (c1) For  $r(t) = 5u(t)$  (step input)
  - (c2) For  $r(t) = 5tu(t)$  (ramp input)
  - (c3) For  $r(t) = 5t^2u(t)$  (parabolic input)

## Example 8 (cont'd)

Solution:

- **(a) Stability**

- Characteristic equation

$$1 + G(s) = 0 \Leftrightarrow s^2 + 70s + 4900 = 0$$

$$\zeta = \frac{1}{2}, \quad \omega_n = 70$$

Since all the coefficients in the first column of Routh array have the same sign, the CL system is **stable**.

- **(b) 2% settling time** ( $\zeta < 1$  : underdamped case)

- Time constant is  $T = \frac{1}{|\text{Re}(\text{pole})|} = \frac{1}{\zeta\omega_n} = \frac{1}{35}$

- 2% settling time is  $4T = \frac{4}{35} \rightarrow \boxed{2\% \text{ settling time} = 0.114}$



# Example 8 (cont'd)

- **(c) Steady-state errors**  $G(s) = \frac{4900}{s(s+70)}$

- **(c1)** For  $5u(t)$  (step)

$$K_p = \lim_{s \rightarrow 0} G(s) = \infty \quad \Rightarrow \quad e_{ss} = \frac{5}{1 + K_p} = 0 \quad \Rightarrow \quad \boxed{e_{ss} = 0}$$

- **(c2)** For  $5tu(t)$  (ramp)

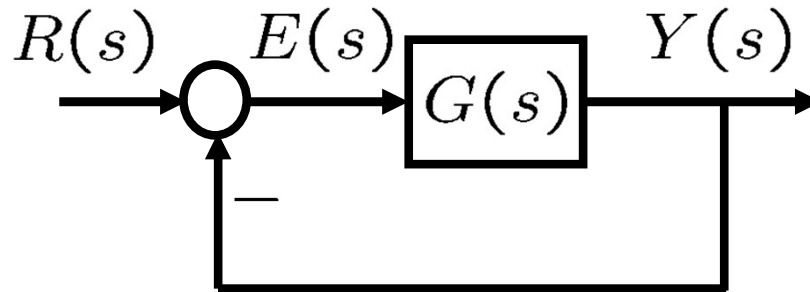
$$K_v = \lim_{s \rightarrow 0} sG(s) = 70 \quad \Rightarrow \quad e_{ss} = \frac{5}{K_v} = \frac{1}{14} \quad \Rightarrow \quad \boxed{e_{ss} = 0.071}$$

- **(c3)** For  $5t^2u(t)$  (parabolic)

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) = 0 \quad \Rightarrow \quad e_{ss} = \frac{10}{K_a} = \infty \quad \Rightarrow \quad \boxed{e_{ss} = \infty}$$

## Example 9

- Design the unity feedback system such that:



$$G(s) = \frac{K(s + \alpha)}{s(s + \beta)}$$

- (a)** Steady-state error for unit ramp input is  $1/10$  and
- (b)** Closed-loop poles are at  $-1 \pm j$

# Example 9

**Solution:**

**Unit ramp input:**  $R \cdot t \cdot u(t)$ ,  $1 \cdot t \cdot u(t) \Rightarrow R = 1$

$$K_v = \lim_{s \rightarrow 0} sG(s) = \lim_{s \rightarrow 0} \frac{sK(s + \alpha)}{s(s + \beta)} = \frac{K\alpha}{\beta} \Rightarrow K_v = \frac{K\alpha}{\beta}$$

$$e_{ss} = \frac{R}{K_v} = \frac{1}{K_v} \stackrel{e_{ss} = \frac{1}{10}}{\Rightarrow} \frac{K\alpha}{\beta} = 10$$

**Find  $\zeta$  and  $\omega_n$ :**

Given pole locations:

$$s_{1,2} = -1 \pm j$$

Comparing with standard form:

$$s_{1,2} = -\zeta\omega_n \pm j\omega_n\sqrt{1 - \zeta^2}$$

Equating real and imaginary parts:

$$-1 = -\zeta\omega_n \Rightarrow \omega_n = \frac{1}{\zeta}$$

$$\omega_n\sqrt{1 - \zeta^2} = 1 \Rightarrow \frac{1}{\zeta}\sqrt{1 - \zeta^2} = 1 \Rightarrow \zeta = \frac{1}{\sqrt{2}} \approx 0.707 \Rightarrow \omega_n = \frac{1}{\zeta} = \sqrt{2} \approx 1.414$$

# Example 9

**Find  $\alpha$ ,  $\beta$ , and  $K$ :**

From the closed-loop transfer function:

$$1 + G(s) = 1 + \frac{K(s + \alpha)}{s(s + \beta)} = 0 \Rightarrow s(s + \beta) + K(s + \alpha) = 0 \Rightarrow s^2 + (\beta + K)s + K\alpha = 0$$

Compare with the standard second-order form:

$$s^2 + 2\zeta\omega_n s + \omega_n^2 = 0$$

Matching coefficients:

$$\begin{cases} s^2 + 2\zeta\omega_n s + \omega_n^2 = 0 \\ s^2 + (\beta + K)s + K\alpha = 0 \end{cases} \quad \Rightarrow \quad \begin{cases} \beta + K = 2\zeta\omega_n \\ K\alpha = \omega_n^2 \end{cases}$$

$$\beta + K = 2 \cdot \frac{1}{\sqrt{2}} \sqrt{2} = 2 \Rightarrow \beta + K = 2$$

$$K\alpha = \omega_n^2 = 2$$

$$\frac{K\alpha}{\beta} = 10 \Rightarrow K\alpha = 10\beta \Rightarrow 10\beta = 2 \Rightarrow \beta = 0.2$$

$$\beta + K = 2 \Rightarrow 0.2 + K = 2 \Rightarrow K = 1.8$$

$$K\alpha = 2 \Rightarrow 1.8\alpha = 2 \Rightarrow \alpha = \frac{2}{1.8} \approx 1.111$$

# Example 9

## Summary:

- (a)

$$K_v = \lim_{s \rightarrow 0} sG(s) = \frac{K\alpha}{\beta} \Rightarrow K_v = \frac{K\alpha}{\beta} ; e_{ss} = \frac{R}{K_v} \xrightarrow{e_{ss} = \frac{1}{10}} K_v = 10 \Rightarrow \frac{K\alpha}{\beta} = 10$$

- (b) Closed-loop poles are at  $-1 \pm j$

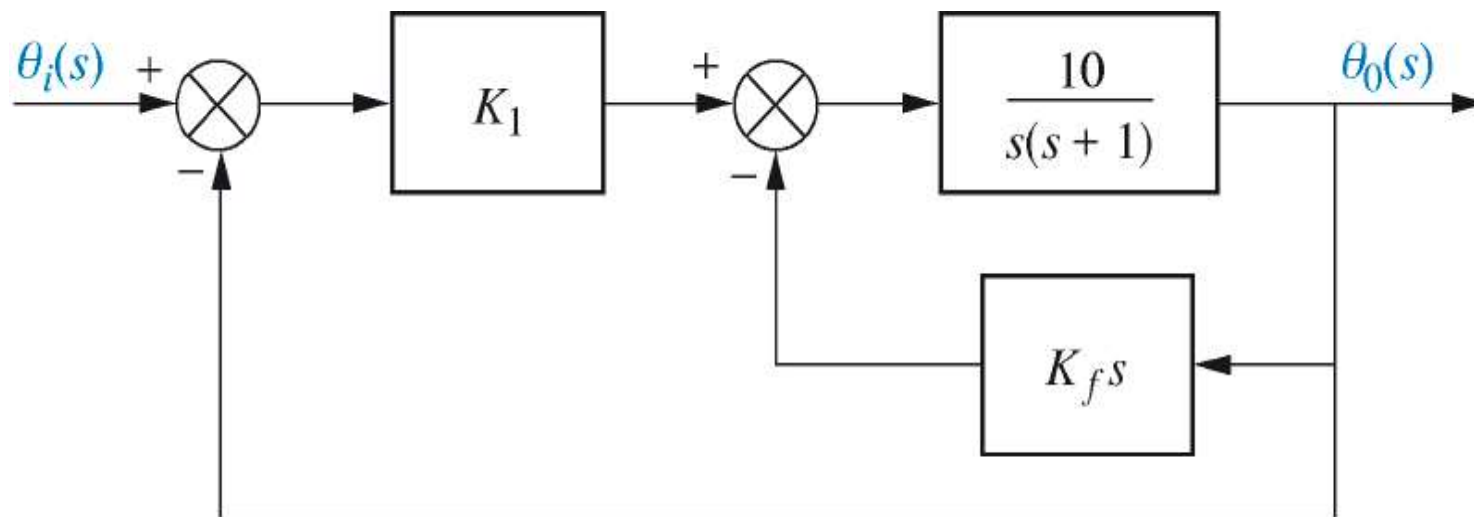
$$\text{characteristic polynomial} = s(s + \beta) + K(s + \alpha) = s^2 + 2\zeta\omega_n s + \omega_n^2 \xrightarrow{s = -\zeta\omega_n \pm j\omega_n\sqrt{1-\zeta^2}}$$

$$\alpha = 1.111, \beta = 0.2, K = 1.8$$

# Example 10

- Design the parameters  $K_1$  &  $K_f$  so that:

$$K_v = 10, \zeta = 0.5$$

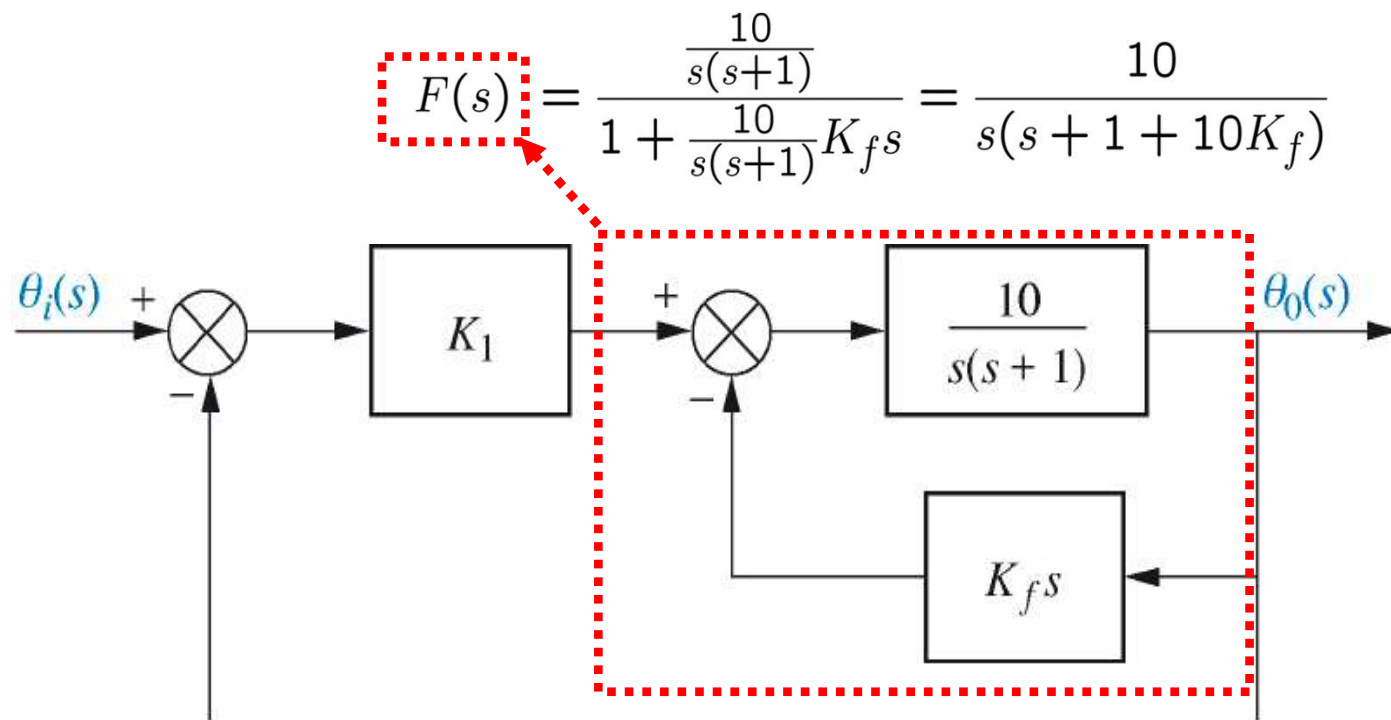


# Example 10 (cont'd)

**Solution:**

- Closed-loop transfer function:

$$\frac{\theta_o(s)}{\theta_i(s)} = \frac{K_1 F(s)}{1 + K_1 F(s)} = \frac{10K_1}{s^2 + (1 + 10K_f)s + 10K_1}$$



## Example 10 (cont'd)

- From the two requirements,

$$K_v = \lim_{s \rightarrow 0} sK_1 F(s) = \frac{10K_1}{1 + 10K_f} = 10 \quad \Rightarrow \quad K_1 = 1 + 10K_f$$

$$s^2 + (1 + 10K_f)s + 10K_1 = s^2 + 2\zeta\omega_n s + \omega_n^2 \quad \Rightarrow \quad \begin{cases} 2\zeta\omega_n = 1 + 10K_f = K_1 \\ \omega_n^2 = 10K_1 \end{cases}$$

$$2\zeta\omega_n = 1 + 10K_f = K_1 \quad \xrightarrow{\zeta = 0.5} \quad \omega_n = K_1 \quad \Rightarrow \quad \text{Substitute for } \omega_n \text{ in } \omega_n^2 = 10K_1$$

$$\Rightarrow K_1^2 - 10K_1 = 0 \quad \Rightarrow \quad \boxed{K_1 = 10, K_f = 0.9}$$



# Summary

- Cruise control example
- Allowable pole locations
- Step responses
  - 2<sup>nd</sup> order overdamped systems
  - Higher order systems
- Design of some feedback systems
- Next
  - Root locus